Gravity Probe B Program P0842 Rev -May 25, 2001

Operations No.

GRAVITY PROBE B PROCEDURE FOR SCIENCE MISSION DEWAR

(PTP) Heat Pulse Meter Verification

P0842 Rev -

May 25, 2001

Prepared by		Checked by	
	Date		_Date
Dave Murray		Jim Maddocks	
Cryogenic Test		Cryogenic Test	
Approvals:			
	Date		Date
Dorrene Ross		Mike Taber	
Quality Assurance		Payload Test Director	
	Date		_Date
Rich Whelan		Robert Brumley	
Systems Engineering		Payload Technical Manage	er

Revision Record

Rev	ECO	Pages	Date
-	-		May 25, 2001

Table on Contents

Α.	SCOPE			1
В.	SAFETY			1
	B.1. Potential Hazards			
	B.2. Mitigation of Hazards			1
	B.3. Injuries			2
C.	QUALITY ASSURANCE			2
	C.1. QA Notification			2
	C.2. Red-line Authority			2
	C.3. Discrepancies			2
D.	TEST PERSONNEL			3
	D.1. Personnel Responsibilities			
	D.2. Personnel Qualifications			3
	D.3. Qualified Personnel			3
E.	REQUIREMENTS			3
	E.1. Electrostatic Discharge Requirements			3
	E.2. Lifting Operation Requirements			3
	E.3. Hardware/Software Requirements			
	E.4. Instrument Pretest Requirements			
	E.5. Configuration Requirements			
	E.6. Optional Non-flight Configurations			/
	E.7. Verification/Success Criteria			
_	E.8. Payload Constraints and Restrictions			
⊦.	REFERENCE DOCUMENTS			
	F.1. Drawings			8
	F.2. Supporting documentationF.3. Additional Procedures			
~				
G.	OPERATIONS			
	G.1. Verify Appropriate QA Notification	ationa Compl	 ata	9
	G.2. Verify Preliminary SMD and GSE Prepar G.3. Verify Configuration Requirements	ations compi	ele	9 Q
	G.4. Record Initial Conditions			
	G.5. Set up Data Acquisition System			-
	G.6. Heat Pulse Heater Checkout			
	G.7. Heat Pulse Operation at 5000 joule - 5 r			
	G.8. Heat Pulse Operation at 11000 joule - 1			
	G.9. Establish Final Configuration			
	G.10. Heat Pulse Meter Data Reduction			15
Η.	PROCEDURE COMPLETION			16
	List of Abbreviation	s and Acrony	vms	
AG-x	Gauge x of Gas Module auxiliary section	MTVC	Main Tank Vent Cap	
AMI	American Magnetics Inc.	MTVC-G	Main Tank Vent Cap pressure ga	
ATC	Advanced Technology Center	MTVC-RV	Main Tank Vent Cap relief valve)
APR-x AV-x	Pressure regulator x of Gas Module Valve x of Gas Module auxiliary section	MTVC-V NBP	Main Tank Vent Cap valve	
CG-x	Gauge x of portable helium pressurization	ONR	Normal boiling point Office of Naval Research	

source Pressure regulator x of portable helium PFCG Fill Cap assembly pressure Gauge

CPR-x

			Page iv
	pressurization source		
CV-x	Valve x of portable helium pressurization source	PFM	Pump equipment Flow Meter
CN [xx] DAS EFM EG-x EM ERV-x EV-x	Data acquisition channel number Data Acquisition System Exhaust gas Flow Meter Gauge x of Gas Module exhaust section Electrical Module Relief valve of Gas Module exhaust section Valve number x of Gas Module exhaust section	PG-x PM psi psig PTD PV-x QA	Gauge x of Pump equipment Pump Module pounds per square inch pounds per square inch gauge Payload Test Director Valve x of the Pump equipment Quality Assurance
FCV FIST GHe GM GP-B GSE GT GTVC-G GTVC-RV GTVC-V GTV-RV GTV-RV GTV-RV GTV-V HX-x KFxx	Fill Cap Valve Full Integrated System Test Gaseous Helium Gas Module Gravity Probe-B Ground Support Equipment Guard Tank Guard Tank Vent Cap Guard Tank Vent Cap pressure gauge Guard Tank Vent Cap relief valve Guard Tank Vent Cap valve Guard Tank vent pressure gauge Guard Tank vent pressure gauge Guard Tank vent relief valve Guard Tank vent valve	RAV-x RGA SMD STV SU SV-x TG-x TV-x UTS Vac VCP-x VCRV-x VCRV-x VCRV-x VCRV-x VCRV-x VCRV-x VCRV-x VCRV-x	Remote Actuated Valve-x Residual Gas Analyzer Science Mission Dewar SMD Thruster vent Valve Stanford University SMD Valve number x Gauge x of Utility Turbo System Valve x of Utility Turbo System Utility Turbo System Vacuum Vent cap pressure gauge Vent cap relief valve Vent cap valve Volts Direct Current Liquid helium Fill line valve Gauge x of Vacuum Module
LHe LHSD LLS LM MT	Liquid Helium Liquid Helium Supply Dewar Liquid level sensor Lockheed Martin Co. Main Tank	VM VV-x VW-x	Vacuum Module Valve x of Vacuum Module Valve x of Dewar Adapter

A. SCOPE

This procedure effects the verification of the heat pulse method of determining the mass of helium in the SMD main tank. This is accomplished by injecting a fixed amount of heat energy into the superfluid helium using internal "heat pulse meter" heaters and measuring the temperature rise of the bulk liquid.

B. SAFETY

B.1. Potential Hazards

Personal injury and hardware damage can result during normal positioning, assembly and disassembly of hardware. Examples include: positioning Dewar in tilt stand; integrating probe with airlock; positioning airlock on Dewar; removing airlock from Dewar; removing probe from Dewar; and positioning support equipment such as pressurized gas cylinders and supply dewars.

A number of undesired events may be associated with these operations. For example, personnel or equipment can be struck when hardware is being moved (e.g. by forklift or crane load). Personnel are subject to entrapment while positioning hardware, such as hands or feet caught between objects as hardware is moved into place. Suspended hardware may be dropped. Personnel can be caught between objects such as forklifts and walls or loads and building support columns.

In addition, liquid helium used in the SMD represents a hazardous material for the personnel involved in the operations. Cryogenic burns can be caused by contact with the cold liquid or gas, high pressures can result if boiling liquid or cold gas is confined without a vent path, and asphyxiation can result if the vent gas is allowed to accumulate.

The SMD Safety Compliance Assessment, document GPB-100153C discusses the safety design, operating requirements and the hazard analysis of the SMD.

B.2. Mitigation of Hazards

B.2.1 Lifting hazards

There are no lifting operations in this procedure

B.2.2 Cryogenic Hazards

The FIST OPS laboratory has an oxygen deficiency monitor that alarms when the oxygen level is reduced to 19.5%. Prior to beginning this procedure in any facility other than the FIST OPS Lab, the presence of a similar oxygen monitor must be verified by safety and operations personnel. Additional temperature and pressure alarms, provided by the DAS, warn of potential over-pressure conditions. Emergency vent line deflectors are installed over the four burst disks on the SMD vacuum shell, and oxygen collection pans are on the floor beneath them.

The following requirements apply to personnel involved in cryogenic operations. Gloves that are impervious to liquid helium and liquid nitrogen are to be worn whenever the possibility of splashing or impingement of high-velocity cryogens exists or when handling equipment that has been cooled to cryogenic temperatures. Protective clothing and full-face shields are to be worn whenever the possibility of splashing cryogens exists.

The FIST Emergency Procedures document, SU/GP-B P0141, discusses emergency procedures. These documents should be reviewed for applicability at any facility where the hardware is operated.

A.1.1. Other Hazards

When appropriate, tools or other items used with the potential to damage the SMD or Probe shall be tethered.

B.3. Injuries

In case of any injury obtain medical treatment as follows LM Call 117; Stanford University Call 9-911

C. QUALITY ASSURANCE

C.1. QA Notification

The ONR representative and SU QA shall be notified 24 hours prior to the start of this procedure. Upon completion of this procedure, the QE Manager will certify his/her concurrence that the effort was performed and accomplished in accordance with the prescribed instructions by signing and dating in the designated place(s) in this document.

C.2. Red-line Authority

Authority to red-line (make minor changes during execution) this procedure is given solely to the PTD or his designate and shall be approved by the QA Representative. Additionally, approval by the Payload Technical Manager shall be required, if in the judgement of the PTD or QA Representative, experiment functionality may be affected.

C.3. Discrepancies

A Quality Assurance Representative designated by D. Ross shall review any discrepancy noted during this procedure, and approve its disposition. Discrepancies will be recorded in a D-log or a DR per Quality Plan P0108. Any time a procedure calls for verification of a specific configuration and that configuration is not the current configuration it represents a discrepancy of one of three types. These types are to be dealt with as described below.

- 1. If the discrepancy has minimal effect on procedure functionality (such as the state of a valve that is irrelevant to performance of the procedure) it shall be documented in the procedure, together with the resolution. Redlines to procedures are included in this category.
- 2. If the discrepancy is minor and affects procedure functionality but not flight hardware fit or function, it shall be recorded in the D-log. Resolution shall be in consultation with the PTD and approved by the QA representative.
- 3. All critical and major discrepancies, those that effect flight hardware fit or functions, shall be documented in a D-log and also in a Discrepancy Report, per P0108.

D. TEST PERSONNEL

D.1. Personnel Responsibilities

The performance of this procedure requires a minimum complement of personnel as determined by the Test Director. The Test Director is the designated signer for the "witnessed by" sign-off located at the end of each procedure. The person in charge of the operation (Test Director or Test Engineer) is to sign the "completed by" sign-off.

D.2. Personnel Qualifications

The Test Director must have a detailed understanding of all procedures and facility operations and experience in all of the SMD operations. Test Engineers must have SMD Cryogenic operations experience and an understanding of the operations and procedures used for the cryogenic servicing/maintenance of the Dewar.

D.3. Qualified Personnel

Test Director	Test Engineer
Mike Taber	Tom Welsh
Dave Murray	Chris Gray
Jim Maddocks	Bruce Clarke
Dave Frank	

E. **REQUIREMENTS**

E.1. Electrostatic Discharge Requirements

This procedure does not involve operations with any equipment sensitive to electrostatic discharge.

E.2. Lifting Operation Requirements

There are no lifting operations in this procedure

E.3. Hardware/Software Requirements

E.3.1 Commercial Test Equipment

No commercial test equipment is required for this operation.

E.3.2 Ground Support Equipment

The Ground Support Equipment includes the Gas Module, the Pump Module, the Electrical Module, and the Vacuum Module. The Gas Module provides the capability to configure vent paths, read pressures and flow rates, and pump and backfill vent lines. The Pump Module provides greater pumping capacity than the Gas Module, together with additional flow metering capabilities. The vent output of the Gas Module flows through the Pump Module. The Electrical Module contains the instruments listed in Table 1 (see the *Electrical Module Manual* for details) and provides remote control of valves in the Gas Module, Pump Module, and SMD. The Vacuum Module contains a turbo pump, backed by a vane pump and provides the capability to pump out the SMD vacuum shell.

This procedure calls for use of hardware located in the Gas Module (Figure 1), the Pump Module (Figure 2), the Vacuum Module (Figure 3), and the Electrical Module (Table 1).

E.3.3 Computers and Software

The Data Acquisition System (DAS) and data acquisition software are required for this procedure. The DAS reads and displays pressures, temperatures, and flow rates and monitors critical parameters. No additional computers or software are required.

E.3.4 Additional Test Equipment

No additional test equipment is required.

E.3.5 Additional Hardware

No additional hardware is required

E.3.6 Tools

No tools are required for this operation.

E.3.7 Expendables

Liquid helium for Guard Tank fillings is required for this operation.

E.4. Instrument Pretest Requirements

The GSE instruments required to perform this procedure are listed in Table 1, together with their serial numbers, where available. Instruments that are required to have current calibrations are indicated in the Cal-Required column. Instruments that do not require calibration are those not used to verify performance requirements and are not connected to flight instrumentation. The status column is to be filled in with the due date of the instrument calibration sticker and verified to be in calibration by QE or QE designee. Serial numbers are to be updated as appropriate.

No.	Loca- Tion				Cal	Status Cal due
		Description	User Name	Serial No.	Required	date
1	DAS	Power Supply, H-P 6627A	A1, A2, A3,A4	3452A01975	Yes	
2	DAS	Power Supply, H-P 6627A	B1, B2, B3, B4	3452A01956	Yes	
3	DAS	Data Acquisition/Control Unit H-P 3497A	-	2936A245539	No	-
4	DAS	Digital Multimeter H-P 3458A	-	2823A15047	Yes	
5	EM	Vacuum Gauge Controller Granville-Phillips Model 316	EG-1a, -1b	2827	No	-
6	EM	Vacuum Gauge Controller Granville-Phillips Model 316	AG-2a, -2b	2826	No	-
7	EM	Vacuum Gauge Controller Granville-Phillips Model 316	EG-3	2828	No	-
8	EM	MKS PDR-C-2C	EG-2, FCG	92022108A	No	-
9	EM	Flow meter – Matheson 8170	EFM-1	96186	No	-
10	EM	Flow meter totalizer Matheson 8124	EFM-1	96174	No	-
11	EM	Liquid Helium Level Controller American Magnetics, Inc. 136	LLS Main Tank	96-409-11	No	-
12	EM	Liquid Helium Level Controller American Magnetics, Inc. 136	LLS Guard Tank	96-409-10	No	-
13	EM	Liquid Helium Level Controller American Magnetics, Inc. 136	LLS Well	96-409-9	No	-
14	EM	Liquid Helium Level Controller American Magnetics, Inc. 136	LLS Axial Lock	96-409-12	No	-
15	EM	Pressure Controller – MKS 152F-92	EV-7a, -7b	96203410A	No	-
16	EM	Power Supply HP 6038A	H08D Tank Heater	96023407A	Yes	
17	EM	Power Supply HP 6038A	H09D Tank Heater	3511A-13332	Yes	
18	EM	Power Supply HP 6038A	RAV Power Supply	3329A-12486	Yes	
19	EM	Vac Ion Pump power supply Varian 929-0910, Minivac	SIP	5004N	No	-
20	EM	Flow meter totalizer Veeder-Root	PFM-1	576013-716	No	-
21	GM	Pressure Gauge, Heise	AG-1	CC-122077	No	-
22	GM	Pressure Gauge, Marshall Town	AG-3	N/A	No	-
23	GМ	Main Tank Heat Exchanger: a) Thermocouple, b) Current meter, c) Temperature set point controller	-	C-19950	No	-
24	GM	Guard Tank Heat Exchanger: a) Thermocouple, b) Current meter, c) Temperature set point controller	-	C-09920	No	-
25	VM	Vacuum Gauge readout, Granville-Phillips 316	VG-3 VG-4	2878	No	-
26	VM	Vacuum Gauge readout, Granville-Phillips 360	VG-1, VG-2 VG-5	96021521	No	-

Table 1. Required Instrumentation and Calibration Status

E.5. Configuration Requirements

E.5.1 Main Tank

The Main Tank liquid must start at SFHe temperatures of 1.7 to 1.85 K or less and be filled to a liquid level \ge 90%. The actuator control valve for the Gas Module valve, EV-9, should be in the "Subatm He" position.

E.5.2 Guard Tank

The Guard Tank may be filled or empty.

E.5.3 Well

The Well may be connected to the UTS via the Well manifold and can be pumped to vacuum conditions

E.5.4 SMD Vacuum Shell

The Vacuum Shell pressure must be less than 5×10^{-5} torr. Procedure P0213, contains the steps for connecting to and pumping on the SMD vacuum shell and should be used it the pressure criterion is not met.

- E.5.5 Alarm System
 - E.5.5.1 The DAS alarm system must be enabled and contain the following alarm set-points:
 - a. Station 200 temperature (CN 01) set at T \leq 2.2 K.
 - b. Top of lead bag temperature set (CN 28) at T \leq 2.2 K.
 - c. Relative Guard Tank Pressure (CN 46) set at $\Delta P \ge 0.3$ torr.
 - E.5.5.2 The Facility Main Alarm System must be armed.
- E.5.6 GSE and Non-flight Hardware
 - E.5.6.1 The ion-pump magnet is installed.
 - E.5.6.2 GSE cabling must be connected between the SMD and the Electrical Module (P/N 5833812) and between the SMD and the Data Acquisition System (P/N 5833811).
 - E.5.6.3 The Main Tank vent line must be connected to the Gas Module with a vacuum insulated line (P/N 5833806).
 Procedures P0674, and P0672 contain the procedures for connecting Main Tank vent lines.
 - E.5.6.4 The Guard Tank vent line must be connected to the Gas Module with a vacuum insulated line (P/N 5833813). Procedure No. P0676 contains the steps for connecting the Guard Tank vent line.
 - E.5.6.5 The high-vacuum pumping line may be connected between the SMD at SV-14 and the inlet port of the Vacuum Module and be pumping up to a closed valve SV-14. Procedure No. P0213 contains the procedure for connecting to and pumping on the SMD vacuum shell.

- E.5.6.6 A relief valve assembly (FLRV) or flight-like burst disk is installed in place of the flight SMD fill-line burst disk (BD3).
- E.5.6.7 The Fill Cap Assembly must be installed at SV-13.
- E.5.6.8 The thruster vent port is flanged to a relief valve/pressure sensor assembly.
- E.5.6.9 The heaters on the SMD top plate, SV-9, and Main Tank vent bayonet must be installed and operational.
- E.5.6.10 The Pump Module must be connected to the Gas module at EV-21/22.

E.6. Optional Non-flight Configurations

The following modifications or non-flight arrangement of the basic SMD configuration may also be in place. They are incidental to the performance of this procedure and not required.

- 1. The SMD is installed in its transportation and test fixture.
- 2. A foreign object and debris shield may cover the upper cone of the SMD and is required whenever work is being performed above the SMD such that hard objects could be dropped and impact the SMD or Probe.

E.7. Verification/Success Criteria

This procedure demonstrates capability of the heat pulse meter to measure the residual superfluid helium in the Main Tank.

Paragraph No.(PLSE-12)	Specification/Req- uirement Title	Criterion
3.7.5.11.1	Heat Pulse Meter	A heat pulse meter shall be provided for measuring the residual superfluid helium in the main tank while in orbit, to within +/- 5% of its true value.

Payload Constraints and Restrictions

- E.7.1 No Payload operations, which result in substantial heat dissipation in the Probe/SMD, e.g., thermal control operations, may be performed during this procedure.
- E.7.2 Vacuum integrity of Probe and Well must be maintained <5 E-6 torr level.

F. **REFERENCE DOCUMENTS**

F.1. Drawings

Drawing No.	Title
LMMS-5833394	Instrumentation Installation

F.2. Supporting documentation

Document No.	Title
LMMC-5835031	GP-B Magnetic Control Plan
GPB-100153C	SMD Safety Compliance Assessment
SU/GP-B P0141	FIST Emergency Procedures
LMSC-P088357	Science Mission Dewar Critical Design Review
SU/GP-B P0108	Quality Plan
LMMS GPB-100333	Science Mission Dewar Failure Effects and Causes Analysis
SU/GP-B P059	GP-B Contamination Control Plan
LMMC F277277v4.2	Science Payload Specification CDRL/PLSE-12

F.3. Additional Procedures

Document No.	Title
SU/GP-B P0672	Connect Main Tank Vent Line to Gas Module – Main Tank Subatmospheric
SU/GP-B P0676	Connect Guard Tank Vent Line to Gas Module
SU/GP-B P0213	Connect Vacuum Module / Pump on SMD Vacuum Shell
SU/GP-B P0214	Stop Pumping on SMD Vacuum Shell / Disconnect Vacuum Module
SU/GP-B P0209B	External Guard Tank Fill

G.

						Page 9
					Operation Number:	
					Date Initiated:	
					Time Initiated:	
OPEF	RATIO	NS				
G.1.	Verif	y Appropri	ate QA N	otification		
		erify SU QA		tified	,	
	D	ate/time	/	, 		
	o V	erify ONR r	represent	ative notified.		
				tified /	,	
G.2.	Verif	y Prelimina	ary SMD a	and GSE Prepa	rations Complete	
	G.2.1	Record C	auard Tar	nk liquid level	%; temperature	[CN24] <u> </u>
	G.2.2	Record N	<i>l</i> lain Tank	liquid level	%; temperature[C	N09] <u>.</u>
	G.2.3	Verify Ma	ain Tank I	iquid level is > 9	0%.	
	G.2.4	Verify rel line.	ief assem	າbly (FLRV) or fli	ght-like burst disk install	ed on SMD fill
G.3.	Verif	y Configur	ation Red	quirements		
	G.3.1	Verify ior	ւ-pump m	nagnet installed.		
	G.3.2		•	g connected betw d Data Acquisitio	ween SMD and Electrica n System.	al Module and
	G.3.3	Verify Ma	ain Tank v	vent line connect	ted to Gas Module.	
	G.3.4	Verify Gu	lard Tank	vent line conne	cted to Gas Module.	
	G.3.5	,	tuator cor He" posit		/-9, located on Gas Mod	lule, set to
	G.3.6	Verify Pu	ımp Modı	le connected to	Gas Module at EV-21/2	2.
	G.3.7	Verify W	ell pressu	ire:		
		G.3.7.1	π We	ell is not being pu	imped:	
			Installec	d", Operations No	e of P0613, "Repump W o , d measured PW-2	Date
		-				

G.3.7.2 π Well is being pumped:

Record PW-2 _____ torr, TG-1 _____ torr.

G.3.8 Verify Fill Cap Assembly is installed at SV-13.

r

Κ

Κ

tor

%

%

- G.3.9 Verify thruster vent port is flanged to a relief valve/pressure sensor assembly.
- G.3.10 Verify Pump Module is connected to the Gas module at EV-21/22.

G.3.11 Ensure DAS alarm system enabled and record set points.

- G.3.11.1 Station 200 temperature ensure CN [01] on DAS alarm list and set to alarm at T \leq 2.2 K. Record set point.
- G.3.11.2 Top of lead bag temperature ensure CN [28] on DAS alarm list and set to alarm at T \leq 2.2 K. Record set point.
- G.3.11.3 **Relative Guard Tank Pressure** ensure CN [46] on DAS alarm list and set to alarm at $\Delta P \ge$ 0.3 torr. Record set point.
- G.3.12 Ensure DAS watchdog timer and alarm enabled.

G.3.13 Ensure liquid-level alarms enabled and record set points.

- G.3.13.1 **Main Tank** ensure liquid-level alarm set \geq 20%. Record set point.
- G.3.13.2 **Guard Tank** ensure liquid-level alarm set ≥ 20%. _____ Record set point.
- G.3.14 Ensure Facility Main Alarm System enabled.

G.3.15 Ensure Top-plate heaters on SMD are connected and on.

G.3.16 Valve configuration:

Open: EV-13, -17;	all other EVs closed
AV-6	All the AVs closed
PV-2, PV-4	All other PVs closed
AP-1 on, PP-1 and PP-2 off	
Optional (if VTH is open)	
Open: VTH	VW-3 closed.
G.3.17 If VTH of Well is open record PW-2	torr.

G.4. Record Initial Conditions

- G.4.1.1 Record initial SMD vacuum shell pressure as follows:
 - a. Turn on Vac-ion pump and record date/time _____/
 - b. Use DAS [Monitor Data] for CN 99.

- c. When value is steady, record pressure (IP) _____ torr.
- d. Verify Vacuum Shell Pressure < 5 x 10⁻⁵ torr. If not, turn off Vac-ion pump and perform procedure P0213, "Pump SMD Vacuum Space with Vacuum Module", to pump out SMD vacuum shell. Record Op No. _____.
- e. Exit [Monitor Data] and collect data with [Set Data Interval] to 5 min.
- f. When data cycle is complete, turn off Vac-ion pump.
- G.4.1.2 Record data in Table 2.

G.5. Set up Data Acquisition System

Note: Refer to Operating Instructions for mechanics of DAS keyboard/mouse operations.

- G.5.1 Configure the DAS for the "Heat Pulse Meter Test" configuration under the [Data Configs] of the DAS program.
- G.5.2 In [Configs], place DAS in fast scan mode.
- G.5.3 Set/verify set the Main Tank and Guard Tank (if appropriate) liquid-level sampling intervals to 60 minutes.
- G.5.4 Note: Liquid level sensor data for entry into data table should be manually sampled.
- G.5.5 Verify plotter is set up to display the following: main tank temperatures on an expanded, ~.01 K full scale; main tank temperature rate of change; and the power to the heat pulse meter heater.
- G.5.6 Record the DAS software and version in use:
- G.5.7 Record the DAS data file in use:
- G.5.8 Set up special data cycle collection to record:

CN09	Main Tank bottom /a	T09D
CN20	Main Tank Top /a	T10D
CN21	Main Tank Top /b	T11D
CN28	Top lead bag /a	T20D
CN40	Top lead bag /c	T22D
CN60	Heater, Main Tank int	ternal /aH01D

G.5.9 Place DAS configuration in normal scan mode and verify by re-running the "Config4m" and recording the value of the scan control coefficient.

Note: CN09 is always in multiple sample (low noise) data acquisition mode.

G.5.10 Set special data cycle time to <0.5 mins.

G.6. Set up Power Supplies

- G.6.1 At the DAS connect/verify connected the B1 power supply output to the H01D heater input and the B2 power supply output to the H02D heater using the three pin audio cables supplied with the DAS.
 - G.6.1.1 Set both power supplies to an "lset" of 0.07 amp (default value).
- G.6.2 Using manual initiations determine the Main Tank liquid level sensor readings:

LLS A (LLD-1) _____ % LLS B (LLD-2) _____ %.

- G.6.3 Set the Main Tank liquid level sensor selector switch to "off".
- G.6.4 Maintain a data log in Table 3 throughout this test.

G.7. Heat Pulse Heater Checkout

Note: The wattage derived in the following is calculated by the DAS using the DMM voltage readings at the voltage leads of the heater and at the 0.1% resistor shunt in the heater current leads.

- G.7.1 Decrease special data cycle time to minimum, ~.25 minutes.
- G.7.2 Set power supplies B1 and B2 to "disable" using "output on/off".
- G.7.3 Set "Vset: of B1 and B2 to 1.0 volt.
- G.7.4 Test heater H01D:
 - G.7.4.1 Comment to DAS "Test 1 volt to H01D".
 - G.7.4.2 For a one minute interval activate and then deactivate power supply B1 using "output on/off" and verify response of power and temperature on special data printout.
- G.7.5 Test heater H02D:
 - G.7.5.1 Change special data to include CN 62, "H02D".
 - G.7.5.2 Comment to DAS "Test 1 volt to H02D".
 - G.7.5.3 For a one minute interval activate and then deactivate power supply B2 using "output on/off" enable and verify response of power and temperature on special data printout.

G.8. Heat Pulse Operation at 5500 joule – 5 mK

Note: Power output may be estimated using 22 ohms for the heat pulse meter heaters.

- G.8.1 Set up power supply B1 with output disabled (use "output on/off") and "Vset" of 15 volts.
- G.8.2 Start the heater operating by depressing the "output on/off" to enable output and record exact time using DAS clock: _____
- G.8.3 At approximately 540 sec (9 minutes) depress "output on/off" to disable the power supply output and record time: ______.

- G.8.4 Wait for the temperatures in the special data to reach steady state (<.002
- G.8.5 K/min) before proceeding to next step, Execute DAS "Get Data".

G.9. Heat Pulse Operation at 11000 joule – 10 mK

Note: Power output may be estimated using 22 ohms for the heat pulse meter heaters.

- G.9.1 Set up power supply B1 with output disabled (use "output on/off")
- G.9.2 Input a "Vset" of 21 volts.
- G.9.3 Start the heater operating by depressing the "output on/off" to enable output and record exact time using DAS clock: ______.
- G.9.4 At approximately 540 sec (9 minutes) depress "output on/off" to disable the power supply output and record time: ______.
- G.9.5 Wait for the temperatures in the special data to reach steady state (<.002 K/min) before proceeding to next step.
- G.9.6 Execute DAS "Get Data".

G.10. Establish Final Configuration G.10.1 Record Date/Time / . G.10.2 Input comment to DAS "End of Heat Pulse Meter Test" G.10.3 Verify valve configuration: Open: EV-13, -17, -7a/-7b; all other EVs closed AV-6 All other AVs closed PV-2, PV-4 All other PVs closed AP-1 on, PP-1 and PP-2 off Optional (if VTH is open) VW-3 closed. Open: VTH G.10.4 If VTH of Well is open record PW-2 torr. G.10.5 Adjust RAV Configuration G.10.5.1 Using RAV log book, verify: Open: RAV-6B, RAV-3 Closed: RAV -1, -2, -5, -6A, -7 G.10.6 Verify/set the DAS data cycle to 15 minutes. G.10.7 Stop Special Data collection using [other menus] [special data], [stop special data]. G.10.8 Set DAS to normal scan using [other menus], [data config], [normal scan]. G.10.9 Verify all operating liquid level sensors have sampling intervals of 10 minutes. G.10.10Ensure DAS alarm enabled and record set points if changed o Thermal conditions substantially unchanged, alarm set points for Station 200 and lead bag unchanged o Thermal conditions substantially changed, temperature alarm points reset as follows: a. Station 200 set point [CN 1] _____ K (≤ 2.2 K) b. Top of Lead Bag set point [CN 28] _____ K (≤ 2.2 K) _____ torr (>0.3 torr) c. Guard Tank, GTVG G.10.11Ensure DAS watchdog timer and alarm enabled. G.10.12Options: π Continue pumping Well under control of P0613B. π Close out pumping Well, perform final sections of P0613B. π Continue with Well manifold: VTH open VW-3 closed under control of

 π Continue with Well locked up: VTH closed; VW-3 closed.

P0613B.

Heat Pulse Meter Data Reduction

- G.10.13Enter the measure mass of liquid for the two cases, using liquid level and analysis of the level sensor versus true volume (see Appendix A) and liquid density.
- G.10.14Using the special data cycle file, integrate the heater power to obtain the energy deposited (in joules) for the two cases.
- G.10.15Use these data to perform the mass calculation from consideration of mass and temperature rise of the liquid and gas Report this analysis in and "S" document "Heat Pulse Meter Performance". Compare the mass measured using the liquid level sensors with the mass calculated using the heat pulse meter. Verify that they agree within 5%.

	Total	Calc	Mass					
		Delta	Delta	Delta	Delta	Mass		
	Energy	T28	T20	T09	Liquid	From level	Mass – Kg	Error - %
	deposite				Level - %	Sensors -		
	d					Kg		
5 mK								
10 mK								

Table 2

H. PROCEDURE COMPLETION

Completed by:	
Witnessed by:	
Date:	
Time:	

Quality Manager	Date
Payload Test Director	Date

Gravity Probe B Program P0842 Rev -Page 17

Table 3a Heat Pulse Meter Data

Date/ Time	Comments	Main Tank LHe level CN101 %	T.V+Atm CN140 torr	Main Tank Press EG-2 CN114 torr	Thruster Vent CN49 torr	EV-7a/7b Position %/%	Vac Ion Pump press CN99 torr	GT press GTV_G CN46 torr	Flow rate PFM-1 CN110 Ll/hr
						/			
						/			
						/			
						/			
						/			
						/			
						/			
						/			
						/			
						/			
						/			
						/			
						/			
						/			
						/			
						/			
						/			
						/			
						/			
						/			

Gravity Probe B Program P0842 Rev -Page 18

Table 1b Heat Pulse Meter Data

Date/ Time	Flow Meter/a T18D [CN 26]	Flow Meter/b T18D [CN 27]	Top of Lead bag [CN 28]	Main Tank Top [CN 20]	MT Bottom T09D [CN09]	GT Bottom [CN 24]	HEX-1 T04D [CN 04]	Slope of MT Bot [CN 141]	Calc(Tave) Slope [CN28]		
	К	К	К	К	К	К	K	К	ĸ	1	

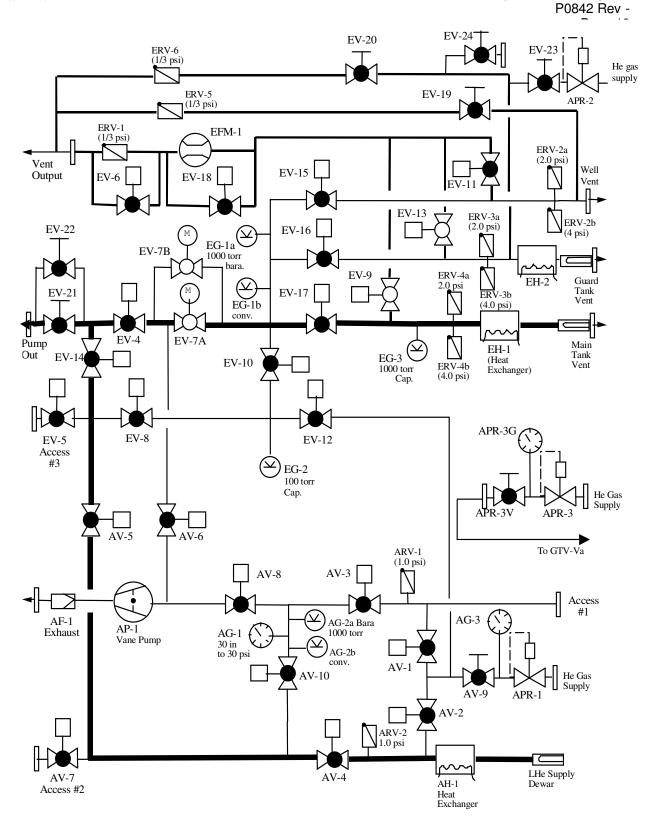


Figure 1. Schematic of Gas Module Plumbing.

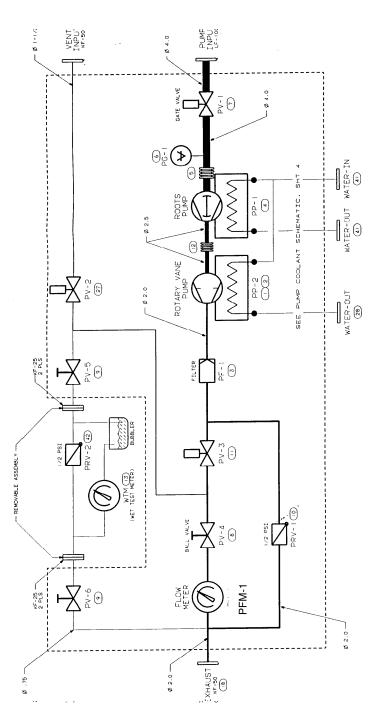
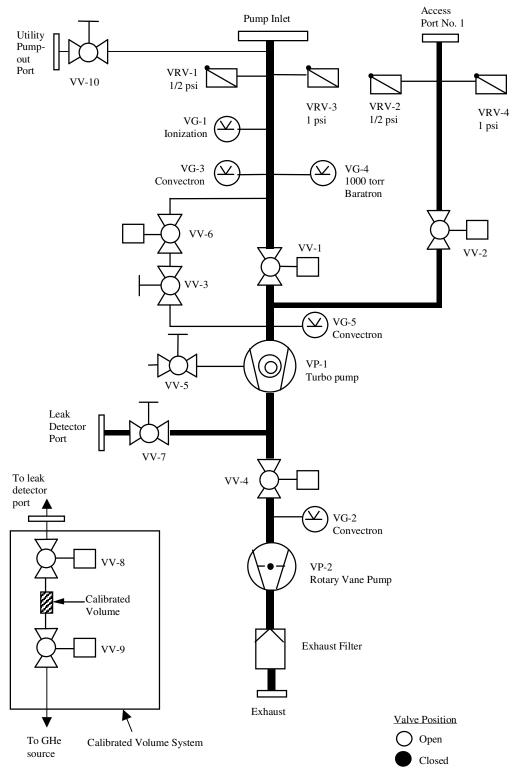


Figure 2. Schematic diagram of Pump Module plumbing.



re 3 Schematic diagram of Vacuum Module

Figu

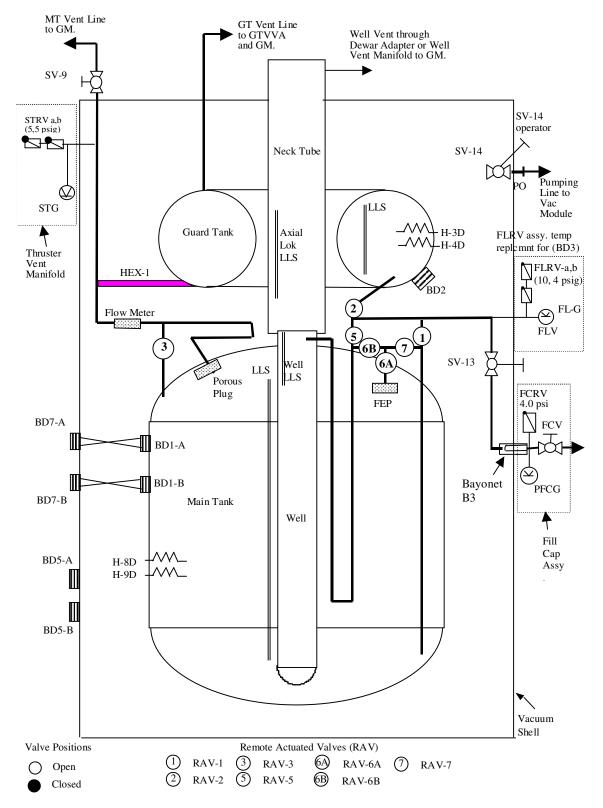


Figure 4. Schematic of Science Mission Dewar plumbing.

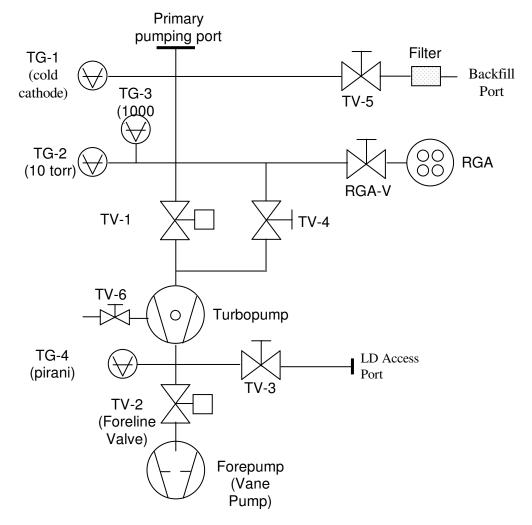


Figure 5. Schematic diagram of Utility Pumping System (UTS)