

# W. W. Hansen Experimental Physics Laboratory STANFORD UNIVERSITY STANFORD, CALIFORNIA 94305 - 4085

## **Gravity Probe B Relativity Mission**

# ATC(ACE) ADP Review - Final Issue Close Out

(Attitude Control Electronics)

S0840, Rev. -

March 28, 2003

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Concurrence:  Systems Engineering	4/24/2003 Date	Concurrence:  Outsity Assurance	Apr. 21,03
Oystems Engineering	Date	Quality Assurance	Date
ITAR Assessment Performed	The !	ITAR Control Req'd?	4.25.03 Vest No
	Tom Langenstein		

## ATC(ACE) ADP Data Review (telecon) and Issues Resolution Meeting Minutes:

Location: Lockheed Martin, Building 255 Room OA235, 1 pm, March 28, 2003

Minutes prepared by: Steve Young

#### References:

LM EM SYS 277, Response to MSFC issues for SPRU, PDU, Battery and SSR, Rich Whelan, March 27, 2003 [note: references are ITAR / U.S. Export Controlled documents]

#### Attendees:

<i>MSFC:</i>	Howard Estes, Albert Froelich, Morris Hammer, Ralph Kissel, Brian Mulac
<i>SU</i> :	Bill Bencze, Dorrene Ross, Steve Young
LM:	Rich Whelan, Mike Sisley, Mike Miranda, Dave Steele

#### Background:

Acceptance Data Packages for all Spacecraft and Space Vehicle components were sent to Marshall Space Flight Center for review. MSFC responded with questions and issues raised by the ADPs. A series of Issue Resolution Meetings were scheduled to address those questions and achieve issue closure based on appropriate discussions, clarifications, or actions. The issue closure process began when several ATC component issues were addressed in January 2003. Meetings scheduled for March and April (2003) will address other subsystems and their components.

The March 28, 2003 meeting focused on the Attitude Control Electronics (ACE), a component of the ATC subsystem.

#### **Overview / Summary:**

Review of the MSFC questions regarding the ACE ADP was divided into two sections. First, those issues not included in EM SYS 277 were discussed and brought to closure pending appropriate clarifications and/or provisions. All issues reviewed and further discussion regarding these items are included below. Next, the ACE component questions included in EM SYS 277 were discussed, along with the "Standard Questions" listed in the referenced document. The complete results of this discussion are not included below. However, several clarifications of the issue resolutions were added and are noted in this document. The remainder of the ACE component issues were closed as stated in the EM.

ACE Questions / Issues from MSFC:

	Sybsystem	MSFC Review	Findings/issues	Finding resolution
		er		
1	Attitude Control Electronics (ACE) (1162- ED-E24761)	Feltner	It appears as though the ADP package submitted for the ACE is actually the ADP representing the effort conducted by SpectrumAstro in response to LM purchase order to remove and replace Interpoint DC/DC converters identified by GIDEP Alert; MSFC No. 7332. The ADP for the original build and test of the ACE was not available or reviewed. Data reviewed: 1162-ED-E24761_I.pdf 2. 1162-ED-E24761_II.pdf 3. 1162-ED-E24761_III.pdf 4. 1162-ED-E24761_IV.pdf 5. 1162-ED-E24761_V.pdf 6. 1162-ED-E24761_V.pdf 7. 8A02482_E Drawing for AFT TIV Bracket 8. LMMS_P086880F.pdf LMMS Specification for the Attitude Control Electronics – March 21, 1999	Mike Miranda (2/17/03) Other ADP Documents have been identified and can be sent to the customer: Top assembly, EMC/EMI, Initial CFT, Post ship AFT, Vibe, Shock.  See clarification below (1) All information is on the VRC. CLOSED
2			1. Although the primary reason for this effort was the removal and replacement of the DC/DC converters it was not clear from the process traveler (0000-EN-K065159a) what the part number/lot date code was for the part removed from location U062 on the two ACI boards impacted. The part number for the replacement part was identified as 5962-9307201HZA but the lot date code was not clear.	withdrawn by MSFC Previous RL answered this Mike Sisley - (3/12/03) The DC/DC converters replaced are on the baseplate (p/n AE- 504836) and are at locations PS001 thru PS004. They all carry a Lot Date Code of 0302. The caps changed were also on the baseplate at locations C102, C106, C202 & C206. They are part number CDR04BX104AKUS with a Lot Date Code of 0043RA.
3			2. There was a discrepancy between the ACI board SNs on the ACE Assembly traveler and the ACI board "as run" post rework test procedures. The ACE assembly traveler identified the SN for the two ACI boards as SN 01480 and SN4. 01481. There were three (?) SN's identified in the ACI board "as run" post rework tests. They were SN 01840, SN 01841, and SN 01842.	withdrawn by MSFC Only 2 ACIs Mike Sisley (3/11/03) RL 574 states there are only two S/N 01840 and 01841.
4			5. There were no parts list (below PWBs) included in the ADP.	Mike Miranda (2/17/03) Build Documents have been sent to the customer (2/11/03): backplane assembly, attitude control interface, TIV, mass trim drive, thruster drive board.  All information is on the VRC. CLOSED
5			6. The Thermal Cycle Test performed was run with one (1) cycle with functional tests only performed at +28 V vs the normal +35V, +28V, and +22V. This precluded as assessment of the linearity of the ACE.	Question references data in rework package see clarification below (1) CLOSED

6		7. The ACE box level TVAC utilized 4 test cycles vs the normal 8 cycles per LM TRR direction. No explanation provided for this change.	Penalty test is 4 cycles only. (3/6/03) Question references data in
			rework package see clarification below (1) CLOSED
7		8. Vol IV of the ACE ADP included an "As Run" test	Mike Miranda - Content was
		procedure for Unpowered Z Axis Vibration test. This	"none" (2/17/03), Mark Anderson
		procedure included a cover sheet for an appendix	- concur (3/6/03)
		titled "Relay Discrepancies" with no contents.	CLOSED
8		9. QAR 11621873 makes mention of questionable	See EM SYS 277
		EGSE performance and references to anomalous	000 Em 010 E/ /
		states with non-conclusive investigations.	<del>.</del>
9		10. At least four (4) out of tolerance conditions were	See EM SYS 277
		explained as caused by test environment noise.	Section 6.2.2 of the EMI test
		QAR 11621815A was one of these. The	procedure (SAI99018) confirms
		investigation report provided with this QAR raises	that the test was performed on
		concern over the ability of the ACE to meet CS01 and CS02 requirements.	the flight unit (not the engineering unit).
10		11. The EMI test procedure and report was not	Mike Miranda (2/17/03)
		found in the ADP although there was reference to a	Other ADP Documents have
		SpectrumAstro document 1079-ET-E01665 titled	been identified and can be sent to
		ACE EMI test procedure found referenced in the	the customer: Top assembly,
	:	ACE TVAC test procedure (1079-ET-E01663).	EMC/EMI, Initial CFT, Post ship
			AFT, Vibe, Shock.
			See clarification below (1)
			All information is on the VRC.
			CLOSED
11		12. The test procedures provided in the ADP make	Mike Miranda (3/10/03)
		reference to a Rev D version of the ACE	Line by line D/F difference
		Specification while the ACE Specification provided, as part of the ADP is a Rev F. Other than the	document ("Changes Between ACE SPEC – P086880 REV D
		Record of Revisions page of LMMS/P086880F it is	and REV F") delivered to MSFC
		not possible to clearly identify the extent of the Rev	review team (4/15/03).
		E and Rev F changes.	CLOSED
12		13. The ACE Specification provided is missing page	
		46. There is no objective evidence of requirements	Mike Miranda - Rev. F rescanned.
13		for an EEE Parts program.	(2/17/03)
13		14. No data was found in the ADP to verify compliance with the radiation requirements. There	See EM SYS 277
		also was no traceability between the ACE	
		Specification radiation requirements and EMI	
		requirements and the Space requirements of	
		P0149.	
14		QAR 11621869 identifies bent pins, which were not	See EM SYS 277
15		replaced.	0 514 03/0 077
13		16. The SpectrumAstro TVAC CFT "as run" procedure (1079-ET-E01661) indicated that the	See EM SYS 277 Question references data in
		following Section 6 tests were not run per Op #530.	rework package see
		a. 6.1 QA Review b. 6.2 Physical Measurements c.	clarification below (1)
		6.3 Enclosure Bonding d. 6.4 Continuity and	- Solow
		Isolation e. 6.5 Setup f. 6.6 40V Input Power g.	
		6.7 Turn On Time h. 6.8 Hot-Swap i. 6.9 Thruster	
		Noise Density j. 6.10 Sixteen Channel Thruster	
		Closed Loop k. 6.11 Telemetry Cross Talk I. 6.12	
		Power Modes m. 6.13 Torque Rod Step Response n. 6.14 In Rush Current	*
		jii. 0. 14 iii nusii Guitelii	

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			o. 6.15 BasePlate Flatness	
			p. 6.16 RAV Current Limit	
lacksquare				
16			Either not found or incomplete: . Drawing Tree	See EM SYS 277
			1. Subsystem, assembly, and subassembly	Question references data in
			hardware (to lowest p/n)	rework package see
			2. Parts procured to a source control drawing	clarification below (1)
			3. Parts procured to a specification control drawing	
			f. List of loose or separate hardware	
			g. Weight and balance Log	
			h. List of authorized deviations and waivers	
			(found only for rebuild effort)	
			j. As designed (Qual) versus as-built (Flight) with	
			explanation of resolution	
			k. Class 1 Discrepancy Reports (found only for	
			rebuild effort)	
			I. Comparison with NASA Alert	
			m. Limited Life Items List	
1			n. As-Measured Mass Properties Report	
			o. Verification Requirements Compliance Document	·
			p. Deviation/Waiver Approval Requests	
17			. Since it is probable that an ADP package for the	Mike Miranda (2/17/03)
			initial delivery of the ACE exists it would probably	Other ADP Documents have
			be prudent to request this from the supplier.	been identified and can be sent to
				the customer: Top assembly,
				EMC/EMI, Initial CFT, Post ship
				AFT, Vibe, Shock.
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				See clarification below (1)
1				All information is on the VRC.
				CLOSED
18			Since the primary reason for the rework effort	
١.,			performed by SpectrumAstro on the ACE was the	withdrawn by MSFC
1			removal and replacement of the Internaliat DO/DO	Mike Sisley - S/N used are 0269,
			removal and replacement of the Interpoint DC/DC	0270, 0271, 0273 (3/11/03)
			converters it would probably be prudent to resolve	
			the SN conflicts identified above to obtain	
			confirmation that the as-built ACE is utilizing the	
<u></u>			desired Interpoint DC/DC converters.	
19			If the Interpoint part number 5962-9307201HZA is	withdrawn by MSFC
			confirmed as utilized in the as-built configuration it	
1			may be prudent to perform a reliability engineering	
			assessment of the risk of any additional thermal	
			cycles on this part before any additional test cycles	
			are performed. The differences between the HZA	· ·
			configuration of this part from the HXC configuration	
			mentioned in the GIDEP alert may not be significant	
			enough to preclude a repeat of the creep problem.	
20	Attitude	Feltner	Comments to follow refer to LM Part # 8A00318. 1.	See EM SYS 277
-	Control	. 5.0101	The parts list and schematics included in the ADP	
	Electronics		do not reflect all design changes and modifications	Delta ADP in delivery process
	(ACE) (1162-			(4/7/03)
	ED-E24761)		that have occurred. Developer is requested to	CLOSED
لـــا	LD-L24/01)		provide updates to documents impacted.	

	Attitude Control Electronics (ACE) (1162- ED-E24761)		2. All box level test data, Items 1 thru 4 above, are no longer valid due to design changes.	See EM SYS 277 Delta ADP in delivery process (4/7/03) CLOSED
	Attitude Control Electronics (ACE) (1162- ED-E24761)	Feltner	3. Selected board level test data are no longer valid and should be resubmitted.	See EM SYS 277 Delta ADP in delivery process (4/7/03) CLOSED
23	Attitude Control Electronics (ACE) (1162- ED-E24761)	Feltner	4. The ACE Specification, Item 21 above, paragraph 3.3.1.1.1 states "derating requirements are not required". This statement requires clarification.	See EM SYS 277 Details of vendor standard practice unavailable parts purchased from qualified aerospace vendor (Spectrum Astro / ISO9000 certified)
	Attitude Control Electronics (ACE) (1162- ED-E24761)	Feltner	5. No data were found in the ADP to verify compliance with the radiation requirements. There also was no traceability between the ACE Specification radiation requirements and EMI requirements and the Space requirements of P0149.	repeat from above See EM SYS 277
	Attitude Control Electronics (ACE) (1162- ED-E24761)	Feltner	6. QAR 11621869 identifies bent pins, which were not replaced. Explain.	repeat from above See EM SYS 277
26	Attitude Control Electronics (ACE) (1162- ED-E24761)	Feltner	7. No Non-Standard Parts Approval Requests (NSPAR) were found in the ADP. Provide.	See EM SYS 277 Mike Miranda (3/25/03) Contained in SAI99017 document on VRC.
27	Attitude Control Electronics (ACE) (1162- ED-E24761)	Feltner	8. The reliability prediction for the ACE required by the specification was not found in the ADP. RL response on this was incomplete.	Drew Costantino (3/11/03) Provided by ACE/PDU Failure Review Action Item #6. (4/13/03) - Data converted to Excel format; delivery to MSFC underway. CLOSED
28	Attitude Control Electronics (ACE) (1162- ED-E24761)	Feltner	The following requirements of Appendix A from Reference 1 were either not found or incomplete: d. Drawing Tree; 1. Parts procured to a source control drawing; 2. Parts procured to a specification control drawing; f. List of loose or separate hardware; g. Weight and balance Log; h. List of authorized deviations and waivers (found only for rebuild effort); j. As designed (Qual) versus as-built (Flight) with explanation of resolution; l. Comparison with NASA Alert; m. Limited Life Items List; n. As-Measured Mass Properties Report; o. Verification Requirements Compliance Document; p. Deviation/Waiver Approval Requests	repeat from above See EM SYS 277

### Additional Notes:

## (1) Response to questions regarding data in rework package, Steve Young (3/31/03)

The first data package provided to MSFC reviewers represented testing performed by the vendor after a rework to remove and replace DC/DC converters. The second data package provided to reviewers represented the original component ADP (prior to the rework) and contains the data in question. Subsequent rework data packages will be provided as available.